

Feedback Control

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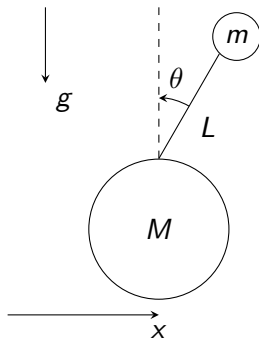
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How do Segways stay up?

Segway



Model



Outline

- ▶ Dynamical systems and stability
- ▶ Controller design (state feedback)
- ▶ Observer design (state estimator)
- ▶ Integral action
- ▶ Simulation

Dynamical Systems

- ▶ Model equations

$$\dot{\mathbf{x}} = f(\mathbf{x}, u) \quad (1)$$

- ▶ Equilibrium

$$f(\mathbf{x}, u) = 0 \quad (2)$$

- ▶ Linearize

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{b}u \quad (3)$$

$$y = \mathbf{C}\mathbf{x} \quad (4)$$

Dynamical Systems - Cart and Pendulum

- ▶ Model equations

$$\ddot{x} = \frac{-mg \sin \theta + mL\dot{\theta}^2 \sin \theta + F}{M + m \sin^2 \theta} \quad (5)$$

$$\ddot{\theta} = \left(\frac{(M+m)g}{L} - m\dot{\theta}^2 \cos \theta - F \frac{\cos \theta}{L \sin \theta} \right) \frac{\sin \theta}{M + m \sin^2 \theta} \quad (6)$$

- ▶ Equilibrium

$$\dot{x} = \ddot{x} = \dot{\theta} = \ddot{\theta} = 0 \Rightarrow \theta = 0 \quad (7)$$

- ▶ Linearize

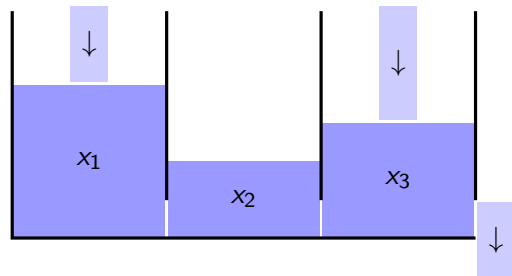
$$\begin{pmatrix} \dot{x} \\ \ddot{x} \\ \dot{\theta} \\ \ddot{\theta} \end{pmatrix} = \begin{pmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & -\frac{mg}{M} & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & -\frac{(M+m)g}{ML} & 0 \end{pmatrix} \begin{pmatrix} x \\ \dot{x} \\ \theta \\ \dot{\theta} \end{pmatrix} + \begin{pmatrix} 0 \\ \frac{1}{M} \\ 0 \\ -\frac{1}{ML} \end{pmatrix} F \quad (8)$$

Dynamical Systems - Three Tanks

Multiple input, multiple output system with hidden state.

$$\begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{pmatrix} = \begin{pmatrix} -1 & 1 & 0 \\ 1 & -2 & 1 \\ 0 & 1 & -2 \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \\ x_3 \end{pmatrix} + \begin{pmatrix} 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{pmatrix} \begin{pmatrix} u_1 \\ u_3 \end{pmatrix}$$

$$\begin{pmatrix} y_1 \\ y_3 \end{pmatrix} = \begin{pmatrix} 1 & 0 & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \\ x_3 \end{pmatrix}$$



Pressure is $\rho g x$, good linear approximation.

Stability

When is an equilibrium \mathbf{x}_0 stable?

- ▶ stable: perturbations dynamically return to \mathbf{x}_0
- ▶ unstable: perturbations dynamically grow away from \mathbf{x}_0



When is a dynamical system stable?

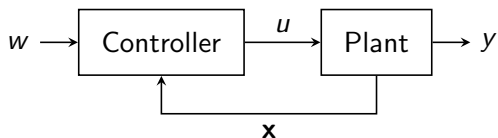
- ▶ Diagonalize the system \rightarrow decoupled dynamics

$$\dot{z}_i = \lambda_i z_i \rightarrow z_i = e^{\lambda_i t} \quad (9)$$

- ▶ System is stable if all eigenvalues are in the **left half plane**

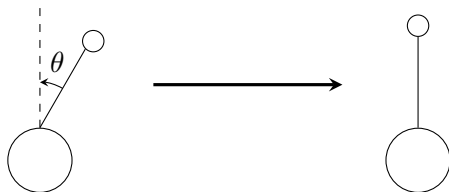
$$\Re(\lambda_i) < 0 \quad (10)$$

Feedback Control

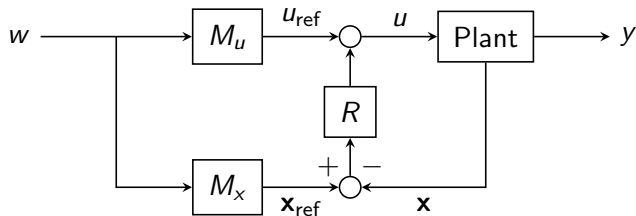


Goals:

- ▶ Stabilize an operating point \mathbf{x}^*
- ▶ Reach \mathbf{x}^* as quickly as possible
- ▶ Reach \mathbf{x}^* with few and small oscillations



Feedback Control



Feedforward: $\mathbf{F} = \mathbf{M}_u + \mathbf{R}\mathbf{M}_x$, Feedback: \mathbf{R}

Open loop

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u$$

$$y = \mathbf{C}^T \mathbf{x}$$

Closed loop

$$\dot{\mathbf{x}} = (\mathbf{A} - \mathbf{B}\mathbf{R})\mathbf{x} + \mathbf{B}\mathbf{F}w$$

$$y = \mathbf{C}^T \mathbf{x}$$

Controllability

Controllability matrix:

$$\mathbf{Q}_c = (\mathbf{B} \quad \mathbf{AB} \quad \mathbf{A}^2\mathbf{B} \quad \dots \quad \mathbf{A}^{n-1}\mathbf{B}) \quad (11)$$

System is **controllable** if \mathbf{Q}_c has full rank n

Controller canonical form:

$$\mathbf{A}_c = \begin{pmatrix} 0 & 1 & 0 & \dots & 0 \\ 0 & 0 & 1 & \dots & 0 \\ 0 & 0 & 0 & \ddots & \vdots \\ 0 & 0 & 0 & 0 & 1 \\ -a_0 & -a_1 & -a_2 & \dots & -a_{n-1} \end{pmatrix} \quad \mathbf{B}_c = \begin{pmatrix} 0 \\ 0 \\ 0 \\ \vdots \\ 1 \end{pmatrix} \quad (12)$$

Controller Canonical Form (CCF)

How can we find the transformation \mathbf{T} ?

Transformation: $\mathbf{A}_c = \mathbf{TAT}^{-1}$, $\mathbf{B}_c = \mathbf{TB}$, $\mathbf{C}_c^\top = \mathbf{C}^\top \mathbf{T}$

$$\begin{aligned} \mathbf{t}_k^\top &= \mathbf{t}_{k-1}^\top \mathbf{A} & \mathbf{t}_k^\top \mathbf{B} &= 0, & \mathbf{t}_n^\top \mathbf{B} &= 1 \\ \mathbf{t}_k^\top &= \mathbf{t}_1^\top \mathbf{A}^{k-1} & \mathbf{t}_1^\top \mathbf{A}^{k-1} \mathbf{B} &= 0, & \mathbf{t}_1^\top \mathbf{A}^{n-1} \mathbf{B} &= 1 \end{aligned}$$

$$\mathbf{t}_1^\top [\mathbf{B}, \mathbf{AB}, \mathbf{A}^2\mathbf{B}, \dots, \mathbf{A}^{n-1}\mathbf{B}] = \mathbf{t}_1^\top \mathbf{Q}_c = \mathbf{e}_n^\top \quad (13)$$

$$\mathbf{TA}_c = \begin{pmatrix} \mathbf{t}_1^\top \\ \mathbf{t}_2^\top \\ \vdots \\ \mathbf{t}_n^\top \end{pmatrix} \begin{pmatrix} 0 & 1 & \dots & 0 \\ 0 & 0 & \ddots & \vdots \\ 0 & 0 & 0 & 1 \\ -a_0 & -a_1 & \dots & -a_{n-1} \end{pmatrix} = \begin{pmatrix} \mathbf{t}_2^\top \\ \mathbf{t}_3^\top \\ \vdots \\ \times \end{pmatrix}$$

$$(\mathbf{t}_1 \quad \mathbf{t}_2 \quad \dots \quad \mathbf{t}_n)^\top \mathbf{B} = (0 \quad 0 \quad \dots \quad 1)^\top$$

Controller Design

- ▶ Pick **desired** closed-loop eigenvalues (left half plane!)

$$\begin{aligned} p(x) &= (x - \lambda_1^*)(x - \lambda_2^*) \cdots (x - \lambda_n^*) \\ &= x^n + p_{n-1}x^{n-1} + \dots + p_1x + p_0 \end{aligned}$$

- ▶ Transform to CCF. Characteristic polynomial is

$$p(x) = x^n + a_{n-1}x^{n-1} + \dots + a_1x + a_0 \quad (14)$$

- ▶ Feedback matrix \mathbf{R}_c (in CCF)

$$r_i = p_i - a_i \quad (15)$$

- ▶ Feedforward \mathbf{M}_x and \mathbf{M}_u : in steady-state, $\dot{\mathbf{x}} = 0$ and $y = w$

$$\begin{pmatrix} \mathbf{A} & \mathbf{B} \\ \mathbf{C}^\top & 0 \end{pmatrix} \begin{pmatrix} \mathbf{M}_x \\ \mathbf{M}_u \end{pmatrix} = \begin{pmatrix} \mathbf{0} \\ \mathbf{I} \end{pmatrix}$$

Example: Controller for Cart and Pendulum

Recall we had the form

$$\mathbf{A} = \begin{pmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & a_{23} & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & a_{43} & 0 \end{pmatrix} \quad \mathbf{B} = \begin{pmatrix} 0 \\ b_2 \\ 0 \\ b_4 \end{pmatrix} \quad (16)$$

Set eigenvalues of $\mathbf{A} - \mathbf{B}\mathbf{R}$ by matching coefficients p_i .

Answer: $\mathbf{R} = (r_0 \quad r_1 \quad r_2 \quad r_3)$

Controllability matrix \mathbf{Q}_c
full rank

$$r_0 = -p_0 ML/g$$

$$r_1 = -p_1 ML/g$$

$$r_2 = -(M+m)g - ML(p_0 L/g + p_2)$$

$$r_3 = -ML(p_1 L/g + p_3)$$

$$\begin{pmatrix} 0 & b_2 & 0 & a_{23}b_4 \\ b_2 & 0 & a_{23}b_4 & 0 \\ 0 & b_4 & 0 & a_{43}b_4 \\ b_4 & 0 & a_{43}b_4 & 0 \end{pmatrix}$$

Linear Quadratic Regulator

If there are multiple inputs, equations for \mathbf{R} underdetermined!

Many solutions! How do we choose?

New goal: get to desired state \mathbf{x}^* with minimum energy.

Minimize a cost function $\mathcal{J}(\mathbf{R})$:

$$\mathcal{J}(\mathbf{R}) = \frac{1}{2} \int_0^{\infty} \mathbf{x}^T(t) \mathbf{Q} \mathbf{x}(t) dt + \frac{1}{2} \int_0^{\infty} \mathbf{u}^T(t) \tilde{\mathbf{Q}} \mathbf{u}(t) dt \quad (17)$$

$\mathbf{Q} \geq 0$ pos semidef; $\tilde{\mathbf{Q}} > 0$ pos def.

These weight the physical states and inputs by **cost factors that we can choose**.

Answer:

$$\mathbf{R} = -\tilde{\mathbf{Q}}^{-1} \mathbf{B}^T \mathbf{P} \quad (18)$$

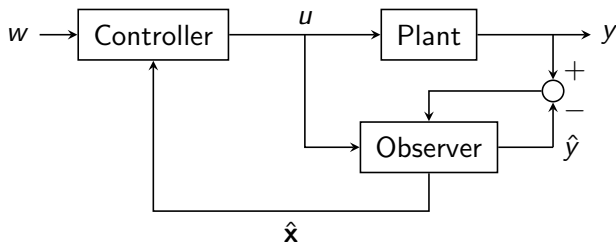
where \mathbf{P} satisfies the Ricatti equation

$$\mathbf{P} \mathbf{B} \tilde{\mathbf{Q}}^{-1} \mathbf{B}^T \mathbf{P} - \mathbf{P} \mathbf{A} - \mathbf{A}^T \mathbf{P} - \mathbf{Q} = 0 \quad (19)$$

Observer

What if we can't measure all the states?

1. Run a simulation of the plant in parallel
2. Compare outputs
3. Feed the difference back to the simulation



$$\begin{aligned}\dot{\hat{x}} &= \mathbf{A}\hat{x} + \mathbf{B}u + \mathbf{L}(y - \hat{y}) & \dot{\hat{x}} - \dot{x} &= \mathbf{A}(\hat{x} - x) + \mathbf{L}(y - \hat{y}) \\ \dot{x} &= \mathbf{A}x + \mathbf{B}u & &= \mathbf{A}(\hat{x} - x) - \mathbf{L}\mathbf{C}^\top(\hat{x} - x)\end{aligned}$$

Controller-Observer Duality

$$\begin{pmatrix} \dot{\mathbf{x}} \\ y \end{pmatrix} = \begin{pmatrix} \mathbf{A} & \mathbf{B} \\ \mathbf{C}^\top & 0 \end{pmatrix} \begin{pmatrix} \mathbf{x} \\ u \end{pmatrix} \quad (20)$$

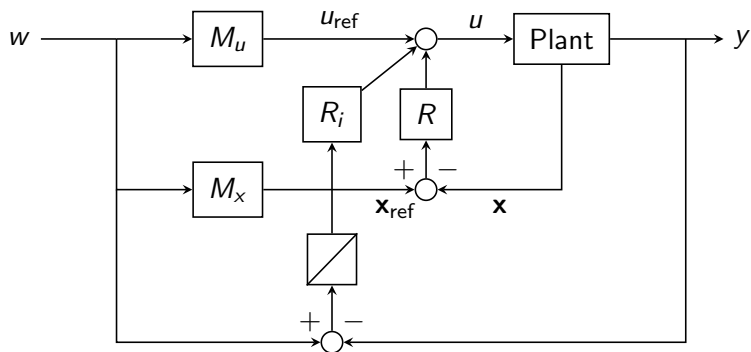
- ▶ Closed loop: $\mathbf{A} - \mathbf{B}\mathbf{R}$
- ▶ Set eigenvalues by choosing \mathbf{R}
- ▶ Controllability matrix \mathbf{Q}_c
 $(\mathbf{B} \quad \mathbf{A}\mathbf{B} \quad \dots \quad \mathbf{A}^{n-1}\mathbf{B})$
- ▶ Closed loop: $\mathbf{A}^\top - \mathbf{C}\mathbf{L}^\top$
- ▶ Set eigenvalues by choosing \mathbf{L}
- ▶ Observability matrix \mathbf{Q}_o
 $(\mathbf{C}^\top \quad \mathbf{C}^\top\mathbf{A}^\top \quad \dots \quad \mathbf{C}^\top(\mathbf{A}^\top)^{n-1})$

New dynamics \rightarrow more eigenvalues.

$$\begin{pmatrix} \dot{\mathbf{x}} \\ \dot{\hat{\mathbf{x}}} \end{pmatrix} = \begin{pmatrix} \mathbf{A} & -\mathbf{B}\mathbf{R} \\ \mathbf{L}\mathbf{C}^\top & \mathbf{A} - \mathbf{L}\mathbf{C}^\top - \mathbf{B}\mathbf{R} \end{pmatrix} \begin{pmatrix} \mathbf{x} \\ \hat{\mathbf{x}} \end{pmatrix} \quad (21)$$

Integral Action

Compensate constant disturbances and model errors.



$$\begin{pmatrix} \dot{\mathbf{x}} \\ \dot{\mathbf{x}}_i \end{pmatrix} = \begin{pmatrix} \mathbf{A} - \mathbf{B}\mathbf{R} & \mathbf{B}\mathbf{R}_i \\ -\mathbf{C}^\top & 0 \end{pmatrix} \begin{pmatrix} \mathbf{x} \\ \mathbf{x}_i \end{pmatrix} + \begin{pmatrix} \mathbf{B}\mathbf{F} \\ \mathbf{I} \end{pmatrix} w \quad (22)$$